# T2R2 東京科学大学 リサーチリポジトリ Science Tokyo Research Repository

# 論文 / 著書情報 Article / Book Information

Title	Experimental investigation of the effects of adhesive thickness on the fracture behavior of structural acrylic adhesive joints under various loading rates
Authors	Yu Sekiguchi, Chiaki Sato
Citation	International Journal of Adhesion and Adhesives, Vol. 105, 102782
Pub. date	2020, 12
DOI	https://dx.doi.org/10.1016/j.ijadhadh.2020.102782
Creative Commons	See next page.
Note	This file is author (final) version.

# License



Creative Commons: CC BY-NC-ND

Experimental investigation of the effects of adhesive thickness on the fracture behavior of structural acrylic adhesive joints under various loading rates

3

1

2

4 Yu Sekiguchi\*, Chiaki Sato

Yokohama, 226-8503, Japan

5

Institute of Innovative Research, Tokyo Institute of Technology, 4259 Nagatsuta-cho, Midoriku

- 6 7
- 8

## 9

### Abstract

10 Designing joints requires a clear understanding of the appropriate thickness of the adhesive used. 11 Structural acrylic adhesives have rarely been studied in terms of their thickness effect on the joint 12 performance. To this end, the fracture resistance of a second-generation acrylic (SGA) adhesive was 13 experimentally investigated by conducting a double cantilever beam (DCB) test. Because the SGA 14 adhesive whitened when plastically deformed, the change of the plastic region with the crack growth 15 was visualized. Therefore, the relationship between the fracture energy and the adhesive thickness was 16 explained in terms of the plastically deformed area. With a thinner adhesive layer, the entire layer was 17 whitened, and a linear relationship was obtained. In this region, the fracture energy increased from 18 approximately 2 kJ/m<sup>2</sup> with an adhesive thickness of 0.2 mm to approximately 4 kJ/m<sup>2</sup> with 0.6 mm 19 thickness. With more increase in the thickness, the fracture energy increased over 8 kJ/m<sup>2</sup>. Increasing 20 the thickness, however, resulted in partial whitening, yielding a non-linear relationship. Moreover, 21 increasing the loading rate changed fracture behavior. At the opening speed of 2.5 m/s, stick-slip crack 22 propagation was observed in any adhesive thickness, and the critical fracture energy dramatically 23 decreased. In contrast, the arrest fracture energy under unstable crack propagation was independent of 24 the loading rate and the adhesive thickness as approximately  $1 \text{ kJ/m}^2$ .

25

26 Keywords: structural acrylic, fracture toughness, impact loading, bond-line thickness

27

28 \* Corresponding author. E-mail address: sekiguchi.y.aa@m.titech.ac.jp (Y. Sekiguchi)

30 1. Introduction

31 Nowadays, adhesive joints are widely used in many industries such as automotive, aerospace, and 32 electronics. Owing to advantages, such as dissimilar material joining, damping improvement, and 33 design flexibility, adhesive bonding technology has a significant potential to improve the functionality 34 of products and is highly in demand. In particular, structural adhesives need to exhibit remarkably high 35 strength and toughness to reliably bond load-bearing structures, and epoxies are often used as the base 36 material of structural adhesives because of their excellent properties. With the development of 37 materials, other types of polymers, such as polyurethane and acrylic resins, have also been applied as 38 adhesives.

39 Second-generation acrylic (SGA) adhesives are two-component structural adhesives that can be rapidly cured at ambient temperatures [1-3]. A free-radical reaction occurs when two components, in 40 41 which an oxidant and a reductant are contained separately, are mixed and the acrylic resin is 42 polymerized. When the acrylic resin in adhesives is modified using an elastomer for toughening, a 43 phase separation with a sea-island structure occurs [4, 5]. Because of the modulus difference between 44 the sea and island areas, micro cracks and cavities are generated when plastically deformed. Thus, the 45 adhesives whiten and fracture in a ductile manner. In addition, micro-fractures inside the adhesive 46 layer inhibit adhesive failure at the interface and the SGA adhesives show superior cohesive failure 47 performance. The SGA adhesives can fill gaps of a few millimeters and can be cured at room 48 temperatures. Therefore, they are useful for on-site bonding, such as in repairing damaged parts and 49 assembling large structures [3, 6, 7]. Moreover, they can be applied to functionally graded adhesive 50 (FGA) joints [8, 9] by taking advantage of their mutability of flexibility.

51 The adhesive thickness is an important design parameter for adhesive joints, and its effects have been 52 widely investigated, particularly on the fracture toughness [10-16]. Typically, the shear strength 53 decreases with increasing adhesive thickness [15, 17]. In contrast, two types of relationships between 54 the fracture energy and the adhesive thickness have been reported [18, 19]. With increasing the 55 adhesive thickness, the fracture energy peaks and decreases to a certain value for brittle adhesives, 56 whereas it increases and plateaus for ductile and tough adhesives [19]. The relationship in the case of 57 brittle adhesives can be explained using the plastic zone diameter [20]. The fracture energy increases 58 linearly when the plastic zone occupies the entire adhesive layer. It decreases and plateaus when the 59 plastic zone diameter is smaller than the bond-line thickness. Conversely, a plateau appears even when 60 the plastic zone occupies the entire adhesive layer in the case of ductile adhesives [19]. Because an

61 accurate thickness control of adhesive layers is a difficult and time-consuming process in 62 manufacturing sites, a clear understanding of how strictly the adhesive thickness must be controlled is 63 important for joint design. The results reported in most studies are primarily for typical structural 64 epoxy adhesives [10-14, 16, 18-20], while some are for polyurethane or silicone adhesives [15, 21-65 23]. The characteristics of adhesives significantly vary with the properties of the base polymers. 66 Therefore, it is important to investigate the thickness effect for each type of adhesive.

- Because adhesives are composed of polymers, they exhibit viscoelasticity, and the loading rate is another factor influencing the fracture toughness [20, 24-26]. In particular, for SGA adhesives, the glass transition temperature (Tg) of acrylic resins, such as polymethyl methacrylate (PMMA), is generally around 100 °C, and it decreases when modified with the elastomer. Thus, many SGA adhesives are sensitive to the operating environment.
- In this study, the fracture behavior of a structural acrylic adhesive was investigated by measuring the mode I fracture energy of the adhesive using double-cantilever beam (DCB) tests. The effects of varying the loading rate, substrate thickness, and adhesive thickness on the fracture toughness were experimentally discussed.
- 76

77 2. Experimental

78 2.1 Specimens and materials

79 Two types of specimens with different substrate thicknesses were manufactured. There were two main 80 reasons to change the substrate thicknesses; (1) to discuss the effect of the substrate rigidity on the 81 plastic zone length at the crack tip and (2) to check the avoidance of the plastic deformation of the 82 substrates. Because the rigidity of the substrate changes the stress state at the crack front, the plastically 83 deformed area of the adhesive layer changes with substrate thickness. Therefore, by comparing the 84 results at different substrate thicknesses, it is possible to determine whether the change in the fracture 85 energy is affected not only by the adhesive thickness but also by the stiffness of the substrate. In 86 addition, it was expected that the fracture energy was extremely increased when the adhesive thickness 87 increased. If the substrate is plastically deformed, or if the experiment deviates significantly from the 88 assumption of the DCB theory, the results at different substrate thicknesses should show totally 89 different dependencies. Therefore, different geometrical specimens were prepared to confirm that it 90 was not a peculiar change due to the shape of the test specimens. Figure 1 shows the specimen 91 geometries, where  $h_{ad}$  is denoted as adhesive thickness. When the fracture energy is the same, the

92 maximum stress applied to the substrate becomes higher with thinner substrate thickness. Therefore, 93 in particular, for thin substrate specimens, it is important to avoid plastic deformation of the substrates, 94 and steel with high yield stress was adopted as the substrate material. However, most of the high-95 strength steels in circulation have a thickness of a few millimeters or less. Therefore, it still had a 96 possibility that the substrate plastically deformed when the adhesive thickness increased. Thus, the 97 adhesively bonded area was narrowed following the method reported in [27]. With this method, it has 98 been confirmed that the fracture energy can be measured in the same manner as the conventional 99 method while the deformation of the substrate can be suppressed and the plastic deformation can be 100 avoided.



Fig. 1. Configuration and dimensions of thin- and thick-substrate DCB specimens. Above: Image ofthe adhesive layer (top view). Bottom: Image of the test specimen after bonded (side view).

104

101

105 The materials used in this study were a two-component acrylic adhesive (Hardloc C355-20 A/B, Denka 106 Co., Ltd., Tokyo, Japan) and steels (carbon steel with the grade S50C for thick substrates and Cr-V-107 based spring steel with the grade SUP10 for thin substrates). The spring steel has a high yield strength 108 of over 1080 MPa, whereas it is over 365 MPa for the S50C steel (values were provided from the 109 manufacturers). The commonly used Young's modulus value for steel (206 GPa) was used for the 110 calculation in this study. The Young's modulus, Poisson's ratio, and Tg of the adhesive were 111 experimentally obtained as 300 MPa, 0.37, and 87 °C, respectively [26, 28]. Figure 2 shows the stress-112 strain relationships of the bulk and cylindrical butt joint specimens [28]. The restraint of the thin 113 adhesive layer by the adherend for the butt joint could be the reason for the high yield stress and the 114 damaged region with stress reduction, which are the major difference from the bulk results.



Fig. 2. Stress-strain diagrams from tensile tests conducted on cylindrical butt joint and bulk specimens[28].

115

The manufacturing process was the same for both the specimen geometries. First, the surfaces of the 119 120 substrate were sandblasted with an Al<sub>2</sub>O<sub>3</sub> grit (SG-118-120 Grid #120, Hozan Tool Ind. Co., Ltd., 121 Osaka, Japan) as an abrasive medium. The sandblasted surface had the surface roughness of Ra  $\approx 1.0$ 122  $\mu$ m, Rz  $\approx$  7.0  $\mu$ m, and RSm  $\approx$  70  $\mu$ m. After the abrasive sand was removed using high-pressure air 123 and wipes, the substrate thickness was measured. The surfaces were wiped with ethanol prior to the 124 bonding. The adhesive thickness was controlled by inserting polytetrafluoroethylene (PTFE) tapes. 125 After the substrates were bonded, they were cured for 24 h at room temperature ( $24\pm 2$  °C). 126 Subsequently, they were placed in an electric furnace at 60 °C for 2.5 h as a post-curing treatment, and 127 the overflowing adhesives were peeled off using an ultrasonic cutter. The specimen thickness was 128 measured after the manufacturing process, and the adhesive thickness was calculated by subtracting 129 the substrate thickness from the overall specimen thickness.

130

131 2.2 DCB tests using universal testing machines

132DCB tests were conducted on the universal testing machine STB-1125S (A&D Co., Ltd., Tokyo,133Japan) with a load cell capacity of 500 N for the thin substrate specimens and on AG-X (Shimadzu134Co., Ltd., Kyoto, Japan) with a load cell capacity of 50 kN for the thick substrate specimens. The crack135length was measured using a CMOS camera installed to observe the specimens from the side. To136deduce the critical fracture energy  $G_c$ , the corrected beam theory (CBT) was applied [29].

137 
$$G_{\rm c} = \frac{3P\delta F}{2b(a+|\Delta|)}$$
(1)

138 where *P* is the load,  $\delta$  is the displacement, *b* is the width of the bonded area, *a* is the crack length, 139 and *F* is the displacement correction factor, which is described as follows [30].

140 
$$F = 1 - \frac{3}{10} \left(\frac{\delta}{a}\right)^2 - \frac{3}{2} \left(\frac{l_1 \delta}{a^2}\right)$$
 (2)

141 where  $l_1$  is the distance from the load point to the substrate axis. In the case of extremely high  $G_c$ 142 value with the thin substrate, the substrate with the non-adhered area largely bend and the length of 143 the cantilever part becomes not equal to the crack length. The correction factor F corrects this discrepancy. The crack length correction  $|\Delta|$  was obtained from the relationship between  $C^{1/3}$  and 144 145 a, where  $C = \delta/P$  is the compliance. In order to avoid the R-curve effect,  $G_c$  values for initiation 146 were not evaluated and the data of  $a > a_0 + 15$  mm, i.e.,  $G_c$  values for propagation, was used for 147 the calculation of stably propagated results. In addition to the crack length measurement, the length 148 from the load point to the whitening front  $a_w$  (see Fig. 3) was optically measured and the whitening 149 length  $a_w - a$  was obtained. Furthermore, the equivalent crack length  $a_e$  was calculated from the 150 load and displacement using the equation

151 
$$a_{\rm e} = \left(\frac{3EI\delta}{2P}\right)^{1/3} (3)$$

152 based on the beam theory, where E is the substrate modulus and I is the moment of inertia of the 153 substrate cross-section. The shear effect on the beam deflection was less than 1% and therefore 154 neglected. The original DCB theory is based upon liner elastic fracture mechanics (LEFM). However, 155 the adhesives largely plastically deform at the crack front. In the case when the adhesive exhibits 156 elasto-plasticity like the SAG adhesives, it has been confirmed by extending the Winkler model and 157 performing elasto-plastic analysis that the fracture energy can be correctly calculated by the CBT [31]. 158 In this case, the crack length correction contains not only the rotation effect of the substrate but also 159 the plastically deformed length of the adhesive layer. Additionally, the equivalent crack length is 160 known to predict the point where the substrate starts bending when the adhesive exhibits elasto-161 plasticity [31]. Therefore, the length  $a_e - a$  also predicts the deformed length of the adhesive layer, 162 including both elastic and plastic deformations, and it is later compared with  $a_w - a$  and  $|\Delta|$ . In the case of an unstable fracture, the fracture energy is determined using the Timoshenko beam theory as 163

164 
$$G = \frac{P^2 F}{3EIb} (3a_e^2 + h^2)$$
 (4)

165 which is the so-called compliance-based beam method (CBBM), where h is the substrate thickness.





167 Figure 3 Schematic of the crack front, whitening front, and equivalent crack front.

169 2.3 DCB tests using a drop weight impact test machine

170 Impact DCB tests were conducted using a drop-weight-type impact testing machine [24]. Specimens

171 with a thin substrate were used. Wedges were set on the weights to open the specimens, as shown in

172 Fig. 4. The displacement and crack length were obtained from the images recorded using a high-speed

173 camera (CRYSTA PI-IPS, Photron Co., Ltd., Tokyo, Japan). Because the load was not recorded during

the impact tests, the load expressed in Eq. (1) was replaced with the displacement and crack length

175 using the simple beam theory. Thus, the fracture energy can be represented as a function of  $\delta$  and a

176 as [26]:

177 
$$G = \frac{9\delta^2 EIF}{4b(a+|\Delta|)^4}$$
 (5)

Because the substrate was thin enough to neglect the kinetic energy of the substrate, the dynamic effecton the fracture energy was not considered.



181 Fig. 4. Image of specimen opening for impact DCB tests [24].

#### 183 3. Results and discussions

184 3.1 DCB tests using universal testing machines

185 Universal testing machines were used to carry out a series of DCB tests on two substrates with different

186 thicknesses, as described previously, by varying the adhesive thickness. In all the configurations, a

187 cohesive fracture was observed.

188 First, quasi-static tests with an obtuse-angle initial crack (type 1, as shown in Fig. 5a) were conducted. 189 The displacement speed was set to 1 mm/min for the thick substrate specimen and 5 mm/min for the 190 thin substrate specimen. Similar relationships between the critical fracture energy and the adhesive 191 thickness were obtained regardless of the substrate thickness, as shown in Fig. 6. The fracture energy 192 increased from approximately 2 kJ/m<sup>2</sup> to over 6 kJ/m<sup>2</sup>. The solid line in the figurer indicates the least-193 square fit of the results for  $h_{ad} < 0.6$  mm, which is also plotted in the other results for reference. The 194 lengths  $a_w - a$  and  $a_e - a$  were different depending on the substrate thickness. The average value 195 of  $a_{\rm w} - a$  was 7.8±1.1 mm for the thin substrate specimen and 29.5±4.3 mm for the thick substrate 196 specimen; the  $a_e - a$  values were 11.2±0.8 and 32.5±4.1 mm, respectively. The length was longer 197 for substrates with a high bending rigidity, but it was more than twice the adhesive thickness even for 198 the thin substrate specimen. Therefore, the process zone in the longitudinal direction was large enough 199 relative to the adhesive thickness. In addition, the equivalent crack front was always a few millimeters 200 ahead of the whitening front. In other words,  $a_e - a_w$  expresses the elastically deformed length of 201 the adhesive layer.



202

Fig. 5. Schematics of initial crack configurations: (a) obtuse angle (type 1), (b) acute angle (type 2),

and (c) V-shape.



Fig. 6. Adhesive fracture energy as a function of the adhesive thickness for an obtuse-angle initialcrack under the quasi-static condition.

205

Because the SGA adhesive was highly ductile, the crack propagated through the center of the adhesive layer in the case of the thin layer even with the obtuse-angle initial crack, as shown in Fig. 7a. With 211 increasing the adhesive thickness, however, the crack position varied and it propagated along with the 212 interface, as shown in Fig. 7b. This change was observed when  $h_{ad}$  was greater than approximately 213 0.7 mm. As a result, the plastic deformation was limited to a part of the adhesive layer. This partial 214 energy consumption led to nonlinearity. Introducing an acute-angle initial crack (type 2, as shown in 215 Fig. 5b) led to a dramatic change in the whitening area for the thicker adhesive layer, as shown in Fig. 216 7c. The linear region expanded to approximately twice the size, as shown in Fig. 8. A clear difference 217 can be observed, particularly for the thick substrate specimens. The fracture energy for type 2 with 218 thicker adhesive layer was increased over 8 kJ/m<sup>2</sup>. However, cracks started propagating once again 219 near the interface when  $h_{ad}$  exceeded approximately 2 mm, and a nonlinear relationship was observed. 220 In contrast, the whitening length in the longitudinal direction was not affected by the adhesive 221 thickness and the crack position, as shown in Fig. 9. Therefore, the whitening occupancy in the 222 thickness direction was an important factor influencing the linear relationship between the fracture 223 energy and the adhesive thickness.

Because the geometry of the thin and thick substrates is different, the amount of plastic energy should be different even if the substrate plastically deforms. Therefore, the existence of plastic deformation can be confirmed by comparing the results. Visually no plastic deformation was observed for both the thin and thick substrates in all cases, but the difference was observed at  $h_{ad} = 3.0$  for type 1, as shown in Fig. 6, which may have a possibility to contain some plastic deformation for the thin substrate. Conversely, no large difference was observed for type 2 comparing the thin and thick results. Therefore, it was confirmed that, in most cases, the plastic deformation was avoided for the tested configuration.







233 crack tip near the interface, and (c) thick adhesive layer with a crack tip at the center.

Fig. 8. Adhesive fracture energy as a function of adhesive thickness under the quasi-static condition with a difference in the initial crack for (a) thick substrate specimen and (b) thin substrate specimen.



238 Fig. 9. Whitening length versus adhesive thickness under quasi-static condition.

237

240 As the loading rate increased, the fracture became unstable, particularly for the thicker adhesive layer. 241 In the case of the thick substrate specimens, unstable crack propagation was observed for most 242 adhesive thicknesses when the displacement speed was increased to the maximum machine speed of 243 0.5 m/min, as shown in Fig. 10a. When the crack propagated unstably, the maximum G value 244 calculated with CBBM was used as the  $G_c$  value for initiation. The strain rate is much higher at crack 245 initiation than during propagation in the case of DCB tests with elastic-plastic adhesives [31]. Thus, 246 the unstable crack propagation may be avoided if the crack is gradually initiated at the beginning. 247 Therefore, a V-shaped initial crack (Fig. 5c) was introduced, and a stable crack propagation could be observed over a wide range of adhesive thicknesses, as shown in Fig. 10b. In addition, the crack 248249 propagation was stable when the displacement speed was slightly decreased to 0.3 m/min, as shown in Fig. 11. At the same time, the deviation between types 1 and 2 became small enough to be 250

251 indistinguishable. A similar change was also observed in the case of the thin substrate specimens. The 252 deviation between types 1 and 2 can hardly be seen when increasing the displacement speed to 0.1 253 m/min, as shown in Fig. 12. With the machine speed increased to a maximum of 1 m/min, unstable 254 crack propagation was observed for the thicker adhesive layer, as shown in Fig. 13. When unstable 255 cracks propagated in the thick substrate specimens, the load decreased as the cracks extended, because 256 the specimen length was too short to permit crack arrest (Fig. 14a). Therefore, only the fracture energy 257 at the beginning of crack initiation was obtained. In contrast, in the thin substrate specimens, the cracks 258 arrested, and stick-slip propagation or initial slip and then stable propagation was observed (Fig. 14b). 259 Thus, the arrest fracture energy  $G_a$  was also obtained.



Fig. 10. Adhesive fracture energy as a function of adhesive thickness for a thick substrate specimenwith a loading rate of 0.5 m/min.



Fig. 11. Adhesive fracture energy as a function of adhesive thickness for a thick substrate specimen

with a loading rate of 0.3 m/min.

263



267 Fig. 12. Adhesive fracture energy as a function of adhesive thickness for a thin substrate specimen

268 with a loading rate of 0.1 m/min.



270 Fig. 13. Adhesive fracture energy as a function of adhesive thickness for a thin substrate specimen



Fig. 14. Load-displacement relationship for (a) thick substrate specimen and (b) thin substrate specimen.

272

276 3.2 DCB tests using an impact test machine

Impact DCB tests were conducted on the thin substrate specimens using a drop-weight impact test machine. The displacement speed was calculated from the images. It was found to be approximately 0.7 m/s when the weight was released from the lowest start position and approximately 2.5 m/s from the highest. In all the configurations, a cohesive fracture was observed.

281 The crack length correction  $|\Delta|$  is essential to calculate the fracture energy in impact DCB tests. 282 Although the load, displacement, and crack length should be measured to obtain  $|\Delta|$  in CBT, the load 283 was not measured in the impact tests in this study. Thus, it cannot be determined directly. From the 284 results obtained using the universal testing machine,  $|\Delta|$  was found to be largely between  $a_e - a$ 285 and  $a_{\rm w} - a$ , as shown in Fig. 15. However, the load was also used to determine  $a_{\rm e}$ . Thus,  $a_{\rm e} - a$ 286 cannot be calculated in the impact tests either. Fortunately,  $a_e - a$  seems to be less independent on 287 the loading rate. Therefore, the average  $a_e - a$  value (10.7 mm) and optically measured  $a_w - a$  at 288 each displacement speed were used to calculate the critical and arrest fracture energies, as shown in

Figs. 16 and 17. The solid line and dashed-dotted line indicate the least-squares fitted lines of  $G_c$ obtained using  $|\Delta| = a_w - a$  and  $|\Delta| = a_e - a$ , respectively. Although the exact value cannot be determined because of the above reasons, the actual fracture energies were expected to be between these values. The dashed line indicates the average value of  $G_a$ . At both the displacement speeds, stick-

293 slip crack propagation was observed.

294



Fig. 15. Mean values of  $a_w - a$ ,  $|\Delta|$  and  $a_e - a$  for the thin substrate specimens at each loading rate.



Fig. 16. Adhesive fracture energy as a function of adhesive thickness for a thin substrate specimen with a loading rate 0.7 m/s calculated using (a)  $|\Delta| = a_w - a$  and (b)  $|\Delta| = a_e - a$ .



Fig. 17. Adhesive fracture energy as a function of adhesive thickness for a thin substrate specimen with a loading rate 2.5 m/s calculated using (a)  $|\Delta| = a_w - a$  and (b)  $|\Delta| = a_e - a$ .

304 In the case of the stick-slip crack propagation, whitening was only generated when the crack was stuck, 305 as shown in Fig. 18a. Because the energy consumed by the plastic deformation was related to the size 306 of the whitened area, the critical fracture energy decreased in accordance with the decrease in the 307 whitening length accompanying the increase in the loading rate (see Fig. 19). In contrast,  $G_a$ 308 maintained a constant value regardless of the test conditions. During unstable crack propagation, no 309 whitening was observed, as shown in Fig. 18b. Thus, a narrow process zone was expected around the 310 crack tip when the crack was arrested, and  $G_a$  was not affected by the adhesive thickness. In addition, 311 the unstable crack velocity was largely constant at approximately 20 m/s, as shown in Fig. 20. 312 Therefore,  $G_a$  was also independent of the loading rate.





314 Fig. 18. Images of stick-slip propagation under impact loading.





Fig. 19. Schematics of the transition of the whitened area under impact loading.



Fig. 20. Determination of the average crack velocity and unstable crack velocity for a stick-slip crack growth from the crack length versus time plot. (Examples shown are for the loading rate and the adhesive thickness for (a) 0.7 m/s and 0.25 mm, (b) 2.5 m/s and 0.22 mm, respectively.)

317

322 3.3 Loading rate effect

323 Figure 21 shows the overall trend in the relationship between the fracture behavior and the adhesive 324 thickness. The relationship was divided into three phases based on the loading rate. With stable crack 325 propagation under quasi-static conditions, linear and nonlinear relationships were observed depending 326 on the presence of plastic deformation in the adhesive layer (first phase). When increasing the loading 327 rate, a transition from stable to unstable crack propagation was observed (second phase). This occurred 328 particularly when the adhesive layer was thicker. Under the impact loading condition, the cracks 329 propagated unstably at all adhesive thicknesses, and  $G_c$  decreased with increasing loading rate, 330 whereas  $G_{a}$  remained constant (third phase). The crack velocity increased with increasing opening 331 displacement speed; however, the unstable crack velocity remained largely constant, as shown in Fig. 332 22. With a further increase in the loading rate, the average crack velocity of the stick-slip propagation 333 could reach the unstable crack velocity. Therefore, another phase is expected when it reaches a much 334 higher opening displacement speed; this remains to be future work. Figure 23 shows the fracture

energy against the opening displacement speed with several selected adhesive thicknesses. Overall,
the fracture energy tends to decrease as the speed increases, but it can be seen that the thicker the
thickness, the greater the change.

338



339

342

340 Fig. 21. Schematic model for explaining the relationship between the adhesive fracture behavior and





343 Fig. 22. Crack velocity versus opening displacement speed.



Fig. 23. Fracture energy versus opening displacement speed for several adhesive thicknesses.

346

344

347 3.4 Fracture energy partition

348 Pardoen et al. [32] proposed the adhesive fracture energy concept, where the fracture energy is divided 349 into two components: the intrinsic work of the fracture associated with the cohesive zone,  $G_0$ , and the 350 additional contribution to the adhesive fracture toughness arising from the far-field plastic dissipation 351 and stored elastic energy within the adhesive layer,  $G_p$ , as  $G_c = G_0 + G_p$ . Additionally,  $G_0$  was 352 assumed to be independent of the adhesive thickness. Thus, the effect of the adhesive thickness on the 353 fracture energy was only included in  $G_p$ . Depending on the type of adhesive, several other approaches 354 have also been considered [14], and determining the concept that matches the results requires a case-355 by-case analysis, particularly depending on the toughening mechanisms of the adhesive.

356 When elastomer-toughened acrylic adhesives are cured, the phase separation generates a random 357 distribution of the elastomer particles in the acrylic resin [4], and an elastomer-rich area and an acrylic-358 rich area are formed [5]. The micro/nanoscale modulus distribution causes a complex and random 359 stress state. Thus, cavities and microcracks are generated, which are considered key to the toughening 360 mechanisms of structural acrylic adhesives. A detailed observation of the fracture surface revealed that 361 the cracks propagated along the brittle part, i.e., the acrylic-rich area, whereas the ductile part, i.e., the 362 elastomer-rich area, around the cracks was largely plastically deformed [33]. Applying Pardoen's 363 concept, the former crack growth is associated with  $G_0$  and the latter plastic deformation is associated 364 with  $G_p$ . For an unstable crack propagation, the plastic deformation effect almost disappears, and the 365 fracture energy that is independent of the loading rate and adhesive thickness becomes dominant. 366 Therefore,  $G_p \approx 0$  and  $G_0 = G_a$ . In contrast, the fracture energy for a stable crack propagation has 367 both effects. However, the fracture energy is asymptotic to  $G_a$  when the adhesive thickness 368 approaches zero because  $G_p$  is zero at zero thickness. Therefore,  $G_0 = G_a$ , i.e.,  $G_p = G_c - G_a$ , is also 369 assumed when the cracks propagate stably. In the case of the adhesive used in this study, load-370 independent fracture energy (fracture energy in arrest) was approximately 1 kJ/m<sup>2</sup> and zero thickness 371 fracture energy (intercept of the least-square fit) was approximately 0.9 kJ/m<sup>2</sup>, and both values were 372 very close. Therefore,  $G_0$  should be around these values and the rest of the energy was dissipated by 373 the plastic deformation around the crack tip.

The slope of the relationship between the fracture energy and the adhesive thickness expresses the change in the energy consumption per volume with increasing adhesive thickness. The slope in the linear region of the quasi-static DCB tests was approximately 4800 kJ/m<sup>3</sup>, which corresponds to  $G_p/h_{ad}$ . From the tensile test results shown in Fig. 2, the energy consumed per volume under plastic deformation, i.e., the area under the stress–strain curve, was found to be approximately 5000 kJ/m<sup>3</sup> for the bulk specimen and approximately 7000 kJ/m<sup>3</sup> for the cylindrical butt joint specimen. A value similar to the energy consumption per volume from the tensile tests confirmed that the increase in the fracture energy with increasing adhesive thickness was mainly related to the plastic energy dissipation in the adhesive layer.

383

384 4. Conclusion

385 Double cantilever beam tests were conducted to investigate the bond-line thickness effect on the 386 adhesive fracture energy of a second-generation acrylic (SGA) adhesive. The adhesive used in the 387 experiment was highly ductile and whitening was observed when plastically deformed. Therefore, the 388 plastically deformed area at the crack front was visualized and a change in the process zone with 389 varying the adhesive thickness, loading rate, and substrate thickness was clarified. Under the quasi-390 static condition, the fracture energy linearly increased with the adhesive thickness when the entire 391 layer was whitened. However, when the thickness was increased, the crack position moved near the 392 interface, partial whitening was observed, and the linearity was lost. Increasing the loading rate, 393 unstable crack propagation was observed for a thicker adhesive layer and transitioned to stick-slip 394 crack propagation at all adhesive thicknesses under the impact loading condition. Furthermore, for the 395 stick-slip crack propagation, a decrease in the whitened area with an increase in the loading rate led to 396 a decrease in the critical fracture energy. In contrast, the arrest fracture energy was independent of the 397 loading rate and adhesive thickness. Moreover, the critical fracture energy was asymptotic to the arrest 398 fracture energy when the adhesive thickness approached zero. Therefore, it was concluded that the 399 change in the fracture energy of the SGA adhesive with the adhesive thickness is mainly attributed to 400 the plastic energy dissipation around the crack tip, and the intrinsic work of fracture remains constant. 401 However, it was also revealed that the whitening area significantly changed with the initial crack 402 condition and substrate rigidity. Therefore, careful discussion is required when analyzing the fracture 403 behavior of actual structures.

404

405 Acknowledgements

This work was based on results obtained from a project commissioned by the New Energy and
Industrial Technology Development Organization (NEDO). We also would like to thank Cactus
Communications for English language editing.

### 410 References

- Allen KW, Harzinikolaou T, ArmstrongKB. A comparison of acrylic adhesives for bonding
   aluminium alloys after using various surface preparation methods. Int J Adhes Adhes
   1984:4(3):133-6. http://doi.org/10.1016/0143-7496(84)90015-0
- 414 2. Dodiuk H, Kenig S. The effect of surface preparation on the performance of acrylic adhesive joints.
- 415 Int J Adhes Adhes 1988;8(3):159-66. http://doi.org/10.1016/0143-7496(88)90094-2
- 3. Suto H, Yoda K, Watanabe J, Yang L. Development of the low-odor and non-flammable secondgeneration acrylic adhesive and its applications. J Adhes Soc Jpn 2012;48(4):127-36.
  http://doi.org/10.11618/adhesion.48.127
- 4. Kamiyama K, Mikuni M, Matsumoto T. Fracture propagation analysis on two component type
  acrylic adhesive joints. Int J Adhes Adhes 2018;83:76-86.
  http://doi.org/10.1016/j.ijadhadh.2018.02.019
- 422 5. Hayashi A, Sekiguchi Y, Sato C. AFM observation of sea-island structure formed by second
  423 generation acrylic adhesive. J Adhes, in press. http://doi.org/10.1080/00218464.2019.1649148
- 424 6. Haraga K, Taguchi K, Yoda K, Nakashima Y. Assembly technique for control panel enclosures
  425 with the combined use of adhesive and rivets and the reduction of energy consumption. Int J Adhes
  426 Adhes 2003;23:371-6. http://doi.org/10.1016/S0143-7496(03)00066-6
- 427 7. Iwata T, Hayashibara H. Durability and flammability evaluation of SGA structural adhesive joints
  428 consisting of a thick adhesive layer for shipbuilding. J Adhes 2019;95(5-7):614-31.
  429 http://doi.org/10.1080/00218464.2019.1581067
- Nakanouchi M, Sato C, Sekiguchi Y, Haraga K, Uno H. Development of application method for
   fabricating functionally graded adhesive joints by two-component acrylic adhesives with different
   elastic moduli. J Adhes 2019;95(5-7):529-42. http://doi.org/10.1080/00218464.2019.1583562
- 433 Sekiguchi Y, Nakanouchi M, Haraga K, Takasaki I, Sato C. Experimental investigation on strength 9. 434 of stepwise tailored single lap adhesive joint using second-generation acrylic adhesive via shear 435 J and low-cycle shear tests. In Adhes Adhes 2019;95:102438. 436 http://doi.org/10.1016/j.ijadhadh.2019.102438
- 437 10. Ji G, Ouyang Z, Li G, Ibekwe S, Pang SS. Effects of adhesive thickness on global and local mode-
- 438 I interfacial fracture of bonded joints. Int J Solids Struct 2010;47:2445-58.
  439 http://doi.org/10.1016/j.ijsolstr.2010.05.006
- 440 11. Da Silva LFM, de Magalhães FACRG, Chaves FJP, de Moura MFSF. Mode II fracture toughness

- 441 of a brittle and a ductile adhesive as a function of the adhesive thickness. J Ades 2010;86:891-905.
  442 http://doi.org/10.1080/00218464.2010.506155
- Marzi S, Biel A, Stigh U. On experimental methods to investigate the effect of layer thickness on
  the fracture behavior of adhesively bonded joints. Int J Adhes Adhes 2011;31:840-50.

445 http://doi.org/10.1016/j.ijadhadh.2011.08.004

- 446 13. Khoo TT, Kim H. Effect of bondline thickness on mixed-mode fracture of adhesively bonded
  447 joints. J Adhes 2011;87:989-1019. http://doi.org/10.1080/00218464.2011.600868
- 448 14. Cooper V, Ivankovic A, Karac A, McAuliffe D, Murphy N. Effects of bonded gap thickness on
  449 the fracture of nano-toughened epoxy adhesive joints. Polymers 2012;53:5540-53.
  450 http://doi.org/10.1016/j.polymer.2012.09.049
- 451 15. Banea MD, da Silva LFM, Campilho RDSG. The effect of adhesive thickness on the mechanical
  452 behavior of a structural polyurethane adhesive. J Adhes 2015;91:331-46.
  453 http://doi.org/10.1080/00218464.2014.903802
- 454 16. Han X, Jin Y, da Silva LFM, Costa M, Wu C. On the effect of adhesive thickness on mode I
  455 fracture energy -an experimental and modelling study using a trapezoidal cohesive zone model. J
  456 Adhes 2020;96(5):490-514. http://doi.org/10.1080/00218464.2019.1601087
- 457 17. Arenas JM, Narbón JJ, Alía C. Optimum adhesive thickness in structural adhesives joints using
  458 statistical techniques based on Weibull distribution. Int J Adhes Adhes 2010;30:160-5.
  459 http://doi.org/10.1016/j.ijadhadh.2009.12.003
- 460 18. Lee DB, Ikeda T, Miyazaki N, Choi NS. Effect of bond thickness on the fracture toughness of
  461 adhesive joints. Trans ASME 2004;126:14-18. http://doi.org/10.1115/1.1631433
- 462 19. Quan D, Murphy N, Ivankovic A. Fracture behavior of epoxy adhesive joints modified with core463 shell rubber nanoparticles. Eng Fract Mech 2017;182:566-76.
  464 http://doi.org/10.1016/j.engfracmech.2017.05.032
- 465 20. Kinloch AJ, Shaw SJ. The fracture resistance of a toughened epoxy adhesive. J Adhes 1981;12:59466 77. http://doi.org/10.1080/00218468108071189
- 467 21. Boutar Y, Naïmi S, Mezlini S, da Silva LFM, Ali MBS. Characterization of aluminium one-
- 468 component polyurethane adhesive joints as a function of bonded thickness for the automotive
- industry: Fracture analysis and behavior. Eng Fract Mech 2017;177:45-60.
  http://doi.org/10.1016/j.engfracmech.2017.03.044
- 471 22. Manterola J, Cabello M, Zurbitu J, Renart J, Turon A, Jumel J, Urresti I. Effect of the width-to-

- thickness ratio on the mode I fracture toughness of flexible bonded joints. Eng Fract Mech
  2019;218:106584. http://doi.org/10.1016/j.engfracmech.2019.106584
- 23. Rosendahl PL, Staudt Y, Odenbreit G, Schneider J, Becker W. Measuring mode I fracture
  properties of thick-layered structural silicone sealants. Int J Adhes Adhes 2019;91:64-71.
  http://doi.org/10.1016/j.ijadhadh.2019.02.012
- 477 24. Yamagata Y, Lu X, Sekiguchi Y, Sato C. Experimental investigation of mode I fracture energy of
  478 adhesively bonded joints under impact loading conditions. Appl Adhes Sci 2017;5:7.
  479 http://doi.org/10.1186/s40563-017-0087-7
- Lißner M, Alabort E, Erice B, Cui H, Blackman BRK, Petrinic N. On the dynamic response of
  adhesively bonded structures. Int J Impact Eng 2020;138:103479.
  http://doi.org/10.1016/j.ijimpeng.2019.103479
- 26. Sekiguchi Y, Yamagata Y, Sato C. Mode I fracture energy of adhesive joints bonded with adhesives
  with different characteristics under quasi-static and impact loading. J Adhes Soc Jpn
  2017;53(10):330-7. http://doi.org/10.11618/adhesion.53.330
- 486 27. Komatsu K, Sekiguchi Y, Ihara R, Tatsumi A, Sato C. Experimental investigation of an adhesive
  487 fracture energy measurement by preventing plastic deformation of substrates in a double
  488 cantilever beam test. J Adhes 2019; 95(10):911-28.
  489 http://doi.org/10.1080/00218464.2018.1451332
- 490 28. Abe N, Sekiguchi Y, Sato C. Parameter identification of material model of toughened adhesive
  491 polymer for elasto-plastic finite element analysis. J Adhes Soc Jpn 2018;54(10):358-66.
  492 http://doi.org/10.11618/adhesion.54.358
- 493 29. Adhesives Determination of the mode 1 adhesive fracture energy of structural adhesive joints
  494 using double cantilever beam and tapered double cantilever beam specimens. ISO 25217:2009.
- 30. Williams JG. The fracture mechanics of delamination tests. J Strain Analysis 1989;24(4):207-13.
  http://doi.org/10.1243/03093247v244207
- 497 31. Sekiguchi Y, Hayashi A, Sato C. Analytical determination of adhesive layer deformation for
- 498 adhesively bonded double cantilever beam test considering elastic-plastic deformation. J Adhes
  499 2020;96(7):647-64. http://doi.org/10.1080/00218464.2018.1489799
- 32. Pardoen T, Ferracin T, Landis CM, Delannay F. Constraint effects in adhesive joint fracture. J
  Mech Phys Solids 2005;53:1951-83. http://doi.org/10.1016/j.jmps.2005.04.009
- 502 33. Kamiyama K, Mikuni M, Matsumoto T, Matsuda S, Kishi H. Crack growth mechanism on SGA

503 adhesive joints. Int J Adhes Adhes 2020;103:102690.

504 http://doi.org/10.1016/j.ijadhadh.2020.102690